



UNIVERSITI TUN HUSSEIN ONN MALAYSIA

**FINAL EXAMINATION
SEMESTER II
SESSION 2016/2017**

COURSE NAME : SOLID MECHANICS
COURSE CODE : BDU 20802
PROGRAM : BACHELOR OF AERONAUTICAL
ENGINEERING TECHNOLOGY
(AIRCRAFT MAINTENANCE)
EXAMINATION DATE : JUNE 2017
DURATION : 2 HOURS
INSTRUCTION : ANSWER **ALL** QUESTIONS IN
PART A AND **TWO (2)**
QUESTIONS IN **PART B**

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THIS PAPER CONTAINS SEVEN (7) PAGES

PART A: ANSWER ALL QUESTIONS

- Q1** (a) A strain gauge is installed in the longitudinal direction on the surface of an aluminium beverage can, demonstrated in **FIGURE Q1**. The radius-to-thickness ratio of the can is 200. When the lid of the can is popped open, the strain changes by $\epsilon_0 = 170 \times 10^{-6}$. Assuming that $E = 70$ GPa, and $\nu = 0.33$, determine the internal pressure, p in the can before it was opened.

(13 marks)

- (b) A thin cylinder 75 mm internal diameter, 250 mm long with wall thick is subjected to an internal pressure of 7 MPa. Determine

- (i) The change in internal diameter,
(ii) The change in length

(12 marks)

- Q2** (a) For the given state of stress, determine the normal and shearing stresses after the element shown in **FIGURE Q2** has been rotated through

- (i) 25° clockwise,
(ii) 10° counterclockwise

(13 marks)

- (b) For the same state of stresses in **FIGURE Q2**, solve the problem using Mohr's circle.

(12 marks)

PART B: ANSWER TWO (2) QUESTIONS ONLY

- Q3** (a) **FIGURE Q3** shows a solid circular rod AD. Determine
- The portion of the shaft in which the maximum shearing stress occurs,
 - The magnitude of that stress
- (12 marks)

- (b) If a 8 mm diameter hole is drilled through each portion of that shaft AD shown in **FIGURE Q3**, evaluate
- The portion of the shaft in which the maximum shearing stress occurs,
 - The magnitude of that stress

(13 marks)

- Q4** **FIGURE Q4** shows at temperature 21°C, a 0.5 mm gap exists between the ends of the aluminium and stainless steel rods. The properties of the rods are given in **TABLE Q4**. At a later time when the temperature has reached 160°C, determine

TABLE Q4

	Aluminium	Stainless steel
Area, A (mm ²)	1806	774
Young Modulus (GPa)	72	190
Thermal Coefficient, α (/°C)	23.9×10^{-6}	17.3×10^{-6}

- (a) The normal stress in the aluminium rod
- (13 marks)
- (b) The change in length of the aluminium rod
- (12 marks)

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FIGURE Q1

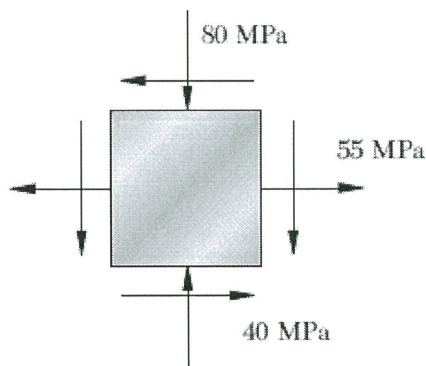


FIGURE Q2

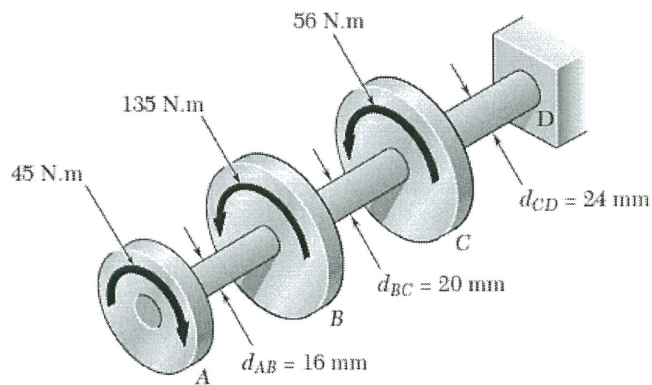


FIGURE Q3

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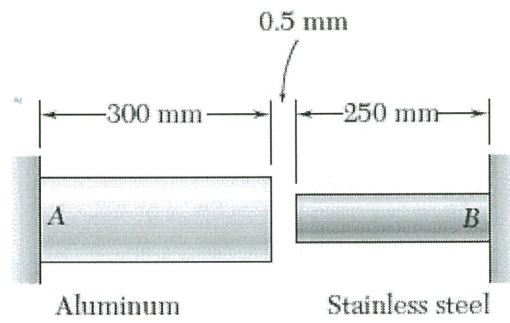


FIGURE Q4

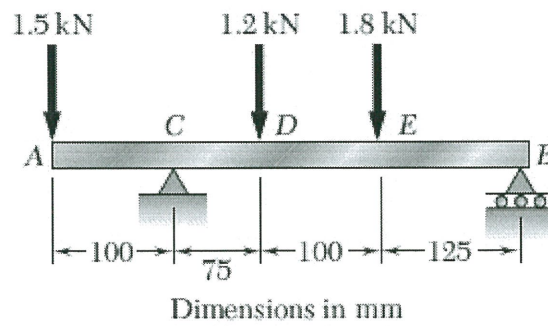


FIGURE Q5

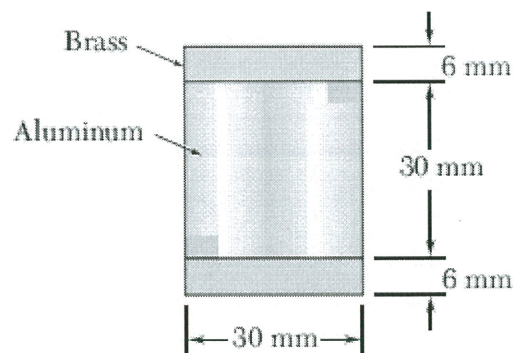


FIGURE Q6

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HASHARU...
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EQUATIONS

$$\sigma_{ave} = \frac{P}{A}$$

$$\delta = \sum_i \frac{P_i L_i}{A_i E_i}$$

$$\delta_T = \alpha(\Delta T)L$$

$$\sigma_{ave} = \frac{\sigma_x + \sigma_y}{2} \quad R = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\sigma_{\max, \min} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\tan 2\theta_p = \frac{2\tau_{xy}}{\sigma_x - \sigma_y}$$

$$\sigma_1 = \frac{pr}{t}$$

$$\sigma_2 = \frac{pr}{2t}$$

$$n = \frac{E_2}{E_1}$$

$$\varepsilon = \frac{1}{E} \pi (\sigma_L - \nu \sigma_H) \quad \tau_{\max} = \frac{Tc}{J} \quad \text{and} \quad \tau = \frac{T\rho}{J}$$

$$J = \frac{1}{2} \pi c^4$$

$$J = \frac{1}{2} \pi (c_2^4 - c_1^4)$$

$$\gamma_{\max} = \frac{\tau_{\max}}{G} = \frac{Tc}{JG}$$

$$\phi = \sum_i \frac{T_i L_i}{J_i G_i}$$

$$T = \frac{P}{\omega} = \frac{P}{2\pi f}$$

$$\tau_{\max} = \frac{T}{c_1 ab^2}$$

$$\phi = \frac{TL}{c_2 ab^3 G}$$

$$\sigma_x = -\frac{My}{I}$$

$$\frac{1}{\rho} = \frac{M}{EI}$$

$$\bar{Y} = \frac{\sum \bar{y}A}{\sum A} \quad I_{x'} = \sum (\bar{I} + Ad^2)$$

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