

UNIVERSITI TUN HUSSEIN ONN MALAYSIA

FINAL EXAMINATION SEMESTER II SESSION 2018/2019

COURSE NAME : SOLID MECHANICS

COURSE CODE : BNJ 20903

PROGRAMME CODE : BNG / BNM / BNK

EXAMINATION DATE : JUNE / JULY 2019

DURATION : 3 HOURS

INSTRUCTION : ANSWER FIVE (5) QUESTIONS

ONLY

THIS QUESTION PAPER CONSISTS OF EIGHT (8) PAGES

Q1	(a)	List ONE (1) example of:			
		(i)	Statically determinate beams.	(2 marks)	
		(ii)	Statically indeterminate beams.	(2 marks)	
	(b)	Figure Q1 (b) shows a beam AB with loads and supports and have an overhan			
		(i)	Determine the support reaction.	(4 marks)	
		(ii)	Sketch the shear diagram.	(5 marks)	
		(iii)	Sketch the moment diagram.	(5 marks)	
		(iv)	Determine the maximum normal stress on a transverse section at C.	(2 marks)	
1750 mm ² . Portion AC is made of a portion BC is made of a high-strength applied at C as shown in Figure Q2 . If			ests of two cylindrical portions AC and BC, each with a cross-section action AC is made of a mild steel with E = 200 GPa and σ_y = 250 made of a high-strength steel with E = 200 GPa and σ_y = 345 MPa. As shown in Figure Q2 . If P is gradually increased from zero until the eaches a maximum value δ_m = 0.3 mm and then decreased back	MPa, and A load P is deflection	
	(a)	The ma	aximum value of P.	(8 marks)	
	(b)	The ma	aximum stress in each portion of the rod.	(6 marks)	
	(c)	The pe	rmanent deflection of C after the load removed.	(6 marks)	

Q3		e Q3 shows a 4 kN.m torque T is applied at end A of the composite shaft. Knowing e modulus of rigidity is 77 GPa for Steel and 27 GPa for Aluminum. Determine:				
	(a)	The n	naximum shearing stress in the steel core.	(7 marks)		
	(b)	The n	naximum shearing stress in the aluminum jacket.	(7 marks)		
	(c)	The a	ngle of twist at A.	(6 marks)		
Q4	(a)	An element of material subjected to plane strain as shown in Figure Q4 (a) has strains as follows: $\epsilon_x = 220 \times 10^{-6}$, $\epsilon_y = 480 \times 10^{-6}$ and $\gamma_{xy} = 180 \times 10^{-6}$. Calculate:				
		(i)	The strains for an element oriented at angle $\theta = 50^{\circ}$	(3 marks)		
		(ii)	Show these strains on a sketch of properly oriented element.	(3 marks)		
	(b)		Determine the equivalent state of stress on an element at the same point in Figure Q4 (b) which represents			
		(i)	The principal stresses.	(3 marks)		
		(ii)	The orientation of principle plane.	(3 marks)		
		(iii)	The maximum in-plane shear stress.	(3 marks)		
		(iv)	Orientation of the plane of maximum in-plane shear stress.	(3 marks)		
		(v)	Average normal stress.	(2 marks)		

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Q5	(a)	Figure Q5 (a) shows the wide-flange section is reinforced with two wooden board. If the composite beam is subjected to an internal moment of $M = 100 \text{ kN.m.}$ Take $E_W = 10 \text{ GPa}$ and $E_{St} = 200 \text{ GPa}$. Determine:					
		(i) The moment of inertia of the transformed section.	(4 marks)				
		(ii) The maximum bending stress of the steel and the wood.	(4 marks)				
	(b)	Figure Q5 (b) shows the wood column has a square cross section with mm length x 100 mm width. It is fixed at its based and free at its to load P that can be applied to the edge of the column without causing fail either by buckling or by yielding. Given $E_w = 12$ GPa and $\sigma_y = 55$	p. Determine the				
Q6		Figure Q6 shows the cantilevered aluminum alloy rectangular beam with $G = 26$ GPa and $E = 68.9$ GPa. Determine:					
	(a)	The internal loadings.	(4 marks)				
	(b)	The shearing strain energy.	(4 marks)				
	(c)	The bending strain energy.	(4 marks)				
	(d)	External work or external force.	(4 marks)				

(e) The conservation of energy.

(4 marks)

-END OF QUESTION-

SEMESTER / SESSION : SEM II / 2018/2019

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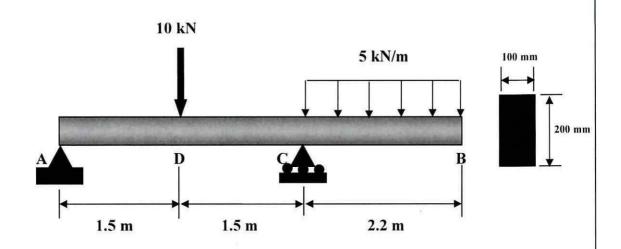


Figure Q1 (b)

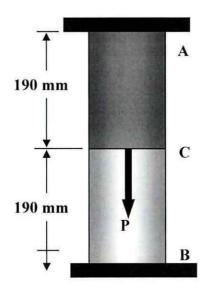


Figure Q2

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: SEM II / 2018/2019

PROGRAMME CODE: BNG/BNM/BNK

COURSE : SOLID MECHANICS

COURSE CODE : BNJ 20903

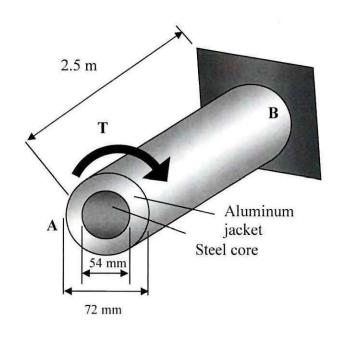


Figure Q3

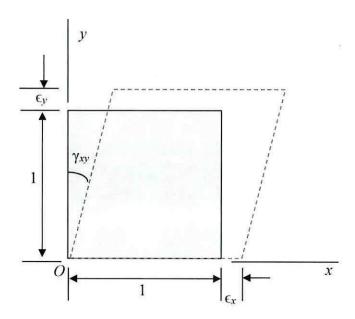


Figure Q4 (a)

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: SOLID MECHANICS

PROGRAMME CODE: BNG/BNM/BNK

COURSE CODE : BNJ 20903

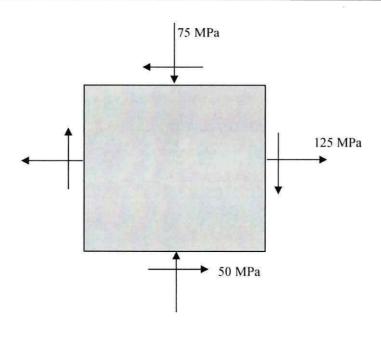


Figure Q4 (b)

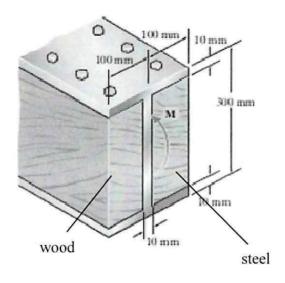


Figure Q5 (a)

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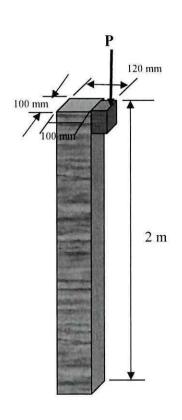


Figure Q5 (b)

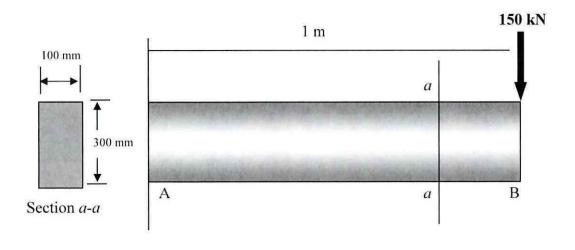


Figure Q6