



UNIVERSITI TUN HUSSEIN ONN MALAYSIA

**FINAL EXAMINATION
SEMESTER I
SESSION 2015/2016**

COURSE NAME : SOLID MECHANICS
COURSE CODE : BNJ 20903
PROGRAMME : 2 BNH/BNK
EXAMINATION DATE : DECEMBER 2015 / JANUARY 2016
DURATION : 3 HOURS
INSTRUCTION : ANSWER FIVE (5) QUESTIONS ONLY

THIS QUESTION PAPER CONSISTS OF **EIGHT (8)** PAGES

- Q1** (a) List **ONE (1)** example of:
- (i) Statically determinate beams (2 marks)
 - (ii) Statically indeterminate beams (2 marks)
- (b) **FIGURE Q1(b)** shows a beam AB with loaded and supported and is said to have an overhang CB. Show all the significant values:
- (i) Determine the support reaction (4 marks)
 - (ii) Sketch the shear diagram (5 marks)
 - (iii) Sketch the moment diagram (5 marks)
 - (iv) Determine the maximum normal stress on a transverse section at C (2 marks)

Q2 Rod AB consists of two cylindrical portions AC and BC, each with a cross-sectional area of 1750 mm^2 . Portion AC is made of a mild steel with $E = 200 \text{ GPa}$ and $\sigma_y = 250 \text{ MPa}$, and portion BC is made of a high-strength steel with $E = 200 \text{ GPa}$ and $\sigma_y = 345 \text{ MPa}$. A load P is applied at C as shown in **FIGURE Q2**. If P is gradually increased from zero until the deflection of point C reaches a maximum value $\delta_m = 0.3 \text{ mm}$ and then decreased back to zero. Determine:

- (a) The maximum value of P (8 marks)
- (b) The maximum stress in each portion of the rod (6 marks)
- (c) The permanent deflection of C after the rod removed (6 marks)

Q3 **FIGURE Q3** shows a 4-kN.m torque T is applied at end A of the composite shaft. Knowing that the modulus of rigidity is 77 GPa for steel and 27 GPa for aluminum. Determine:

- (a) The maximum shearing stress in the steel core (7 marks)
- (b) The maximum shearing stress in the aluminum jacket (7 marks)
- (c) The angle of twist at A (6 marks)

Q4 (a) An element of material subjected to plane strain as shown in **FIGURE Q4(a)** has strains as follows: $\epsilon_x = 220 \times 10^{-6}$, $\epsilon_y = 480 \times 10^{-6}$ and $\gamma_{xy} = 180 \times 10^{-6}$. Calculate:

- (i) the strains for an element oriented at angle $\theta = 50^\circ$ (3 marks)
- (ii) show these strains on a sketch of properly oriented element. (3 marks)

(b) Determine the equivalent state of stress on an element at the same point in **FIGURE Q4(b)** which represents

- (i) the Principal Stresses (3 marks)
- (ii) the Orientation of Principle Plane (3 marks)
- (iii) the Maximum In-Plane Shear Stress (3 marks)
- (iv) Orientation of the Plane of Maximum In-Plane Shear Stress (3 marks)
- (v) Average Normal Stress (2 marks)

Q5 (a) **FIGURE Q5(a)** show the wide-flange section is reinforced with two wooden board. If this composite beam is subjected to an internal moment of $M = 100 \text{ kN.m}$. Take $E_w = 10 \text{ GPa}$ and $E_{st} = 200 \text{ GPa}$. Determine

(i) the moment of inertia of the transformed section (4 marks)

(ii) the maximum bending stress of the steel and the wood (4 marks)

(b) **FIGURE Q5(b)** show the wood column has a square cross section with dimensions 100 mm by 100 mm. It is fixed at its based and free at its top. Determine the load P that can be applied to the edge of the column without causing the column to fail either by buckling or by yielding. Given $E_w = 12 \text{ GPa}$ and $\sigma_y = 55 \text{ MPa}$ (12 marks)

Q6 **FIGURE Q6** show the cantilevered aluminum alloy rectangular beam with $G = 26 \text{ GPa}$ and $E = 68.9 \text{ GPa}$. Determine:

(a) the Internal loadings (4 marks)

(b) the Shearing Strain Energy (4 marks)

(c) the Bending Strain Energy (4 marks)

(d) external work or external force (4 marks)

(e) the Conservation of Energy (4 marks)

-END OF QUESTION-

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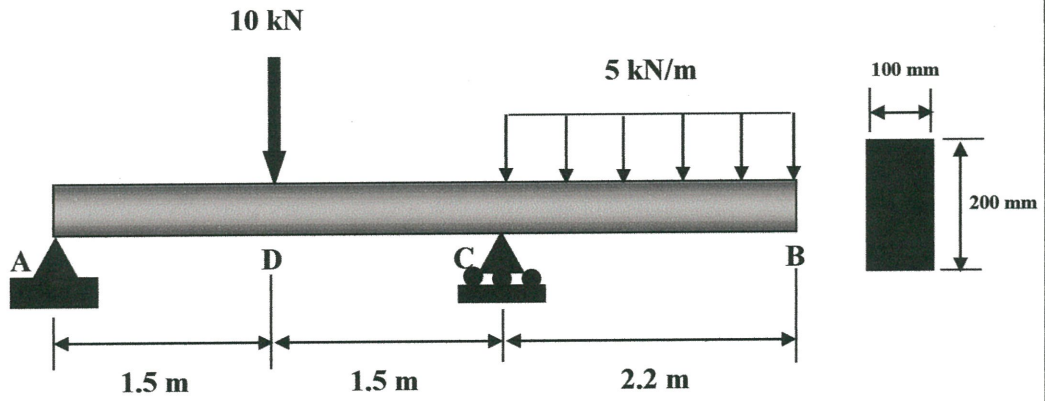


FIGURE Q1(b)

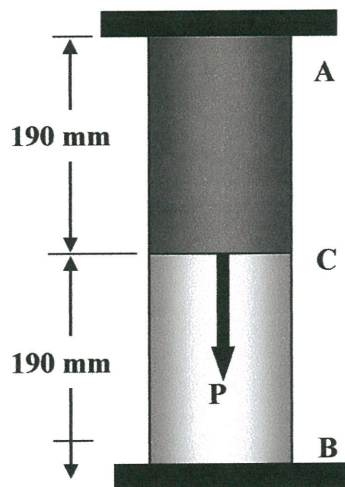


FIGURE Q2

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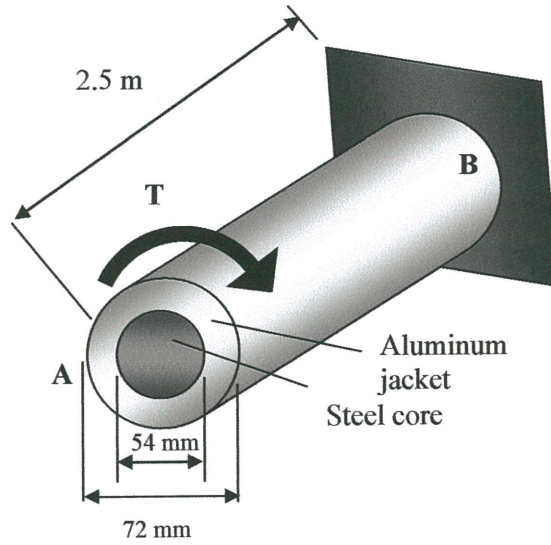


FIGURE Q3

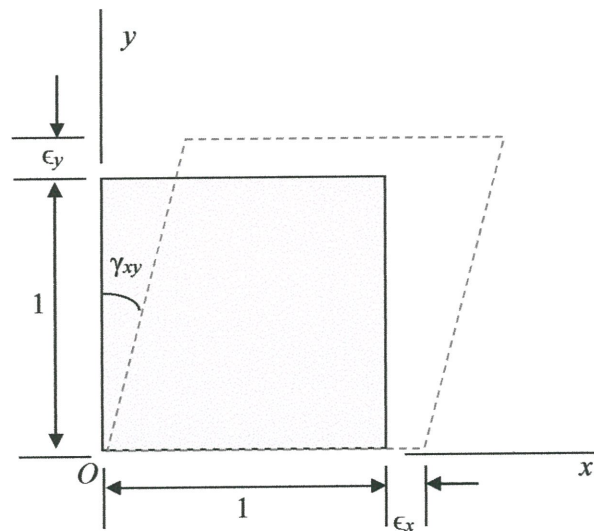


FIGURE Q4(a)

DR NORAH BINTI MARSI
Lecturer
Faculty of Mechanical Engineering
Universiti Teknikal Malaysia Melaka
76100 Durian Tunggal, Melaka

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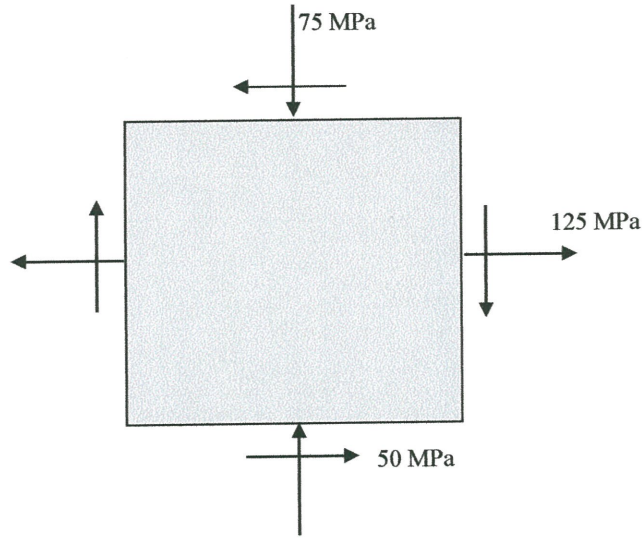


FIGURE Q4(b)

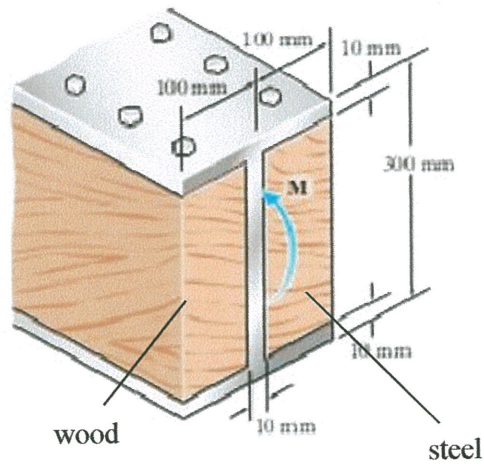


FIGURE Q5(a)

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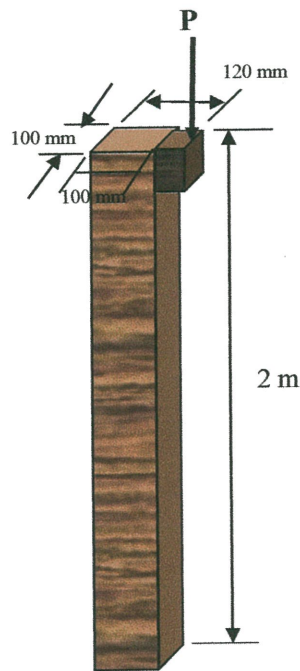


FIGURE Q5(b)

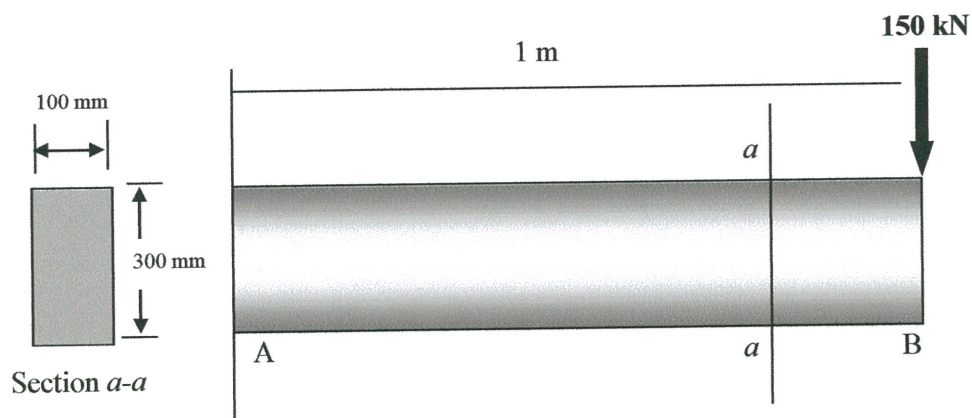


FIGURE Q6