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# UTHM

Universiti Tun Hussein Onn Malaysia

**UNIVERSITI TUN HUSSEIN ONN MALAYSIA**

**FINAL EXAMINATION  
SEMESTER II  
SESSION 2023/2024**

- COURSE NAME : MECHANICS OF MATERIALS
- COURSE CODE : BFC 20903
- PROGRAMME CODE : BFF
- EXAMINATION DATE : JULY 2024
- DURATION : 3 HOURS
- INSTRUCTION :
1. ANSWER ALL QUESTION IN **PART A** AND THREE (3) QUESTION IN **PART B**.
  2. THIS FINAL EXAMINATION IS CONDUCTED VIA  
 Open Book  
 **Closed Book**
  3. STUDENTS ARE **PROHIBITED** TO CONSULT THEIR OWN MATERIAL OR ANY EXTERNAL RESOURCES DURING THE EXAMINATION CONDUCTED VIA CLOSED BOOK

THIS QUESTION PAPER CONSISTS OF **EIGHT (8)** PAGES

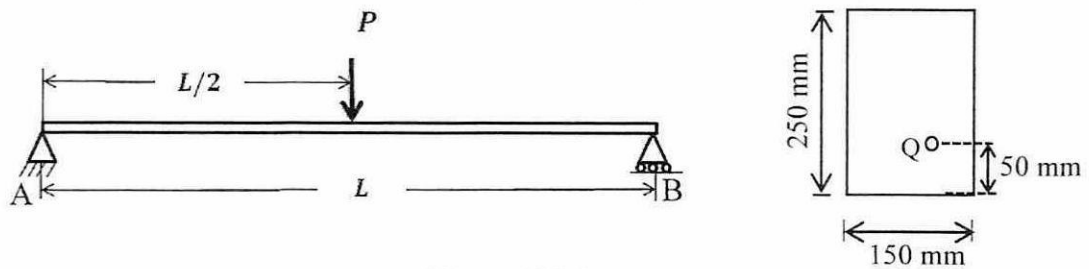
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**PART A - ANSWER ALL QUESTION**

**Q1** Figure Q1.1 shows a plain rectangular beam subjected to point load of 10 kN. The beam is 6 meter long with size of  $b = 150$  mm and  $h = 250$  mm.

- (a) Calculate support reaction at A and B. (5 marks)
- (b) Determine maximum shear,  $V_{max}$  and maximum moment,  $M_{max}$  using shear force and bending moment diagram. (10 marks)
- (c) Calculate maximum bending stress,  $\sigma$  at tension and compression in  $N/mm^2$ . Draw the bending stress diagram. (10 marks)
- (d) Determine bending stress at 75 mm above the neutral axis. (5 marks)
- (e) At support B, compute the shear stress,  $\tau$  at point Q and maximum shear stress,  $\tau_{max}$ . Draw the shear stress diagram. (10 marks)



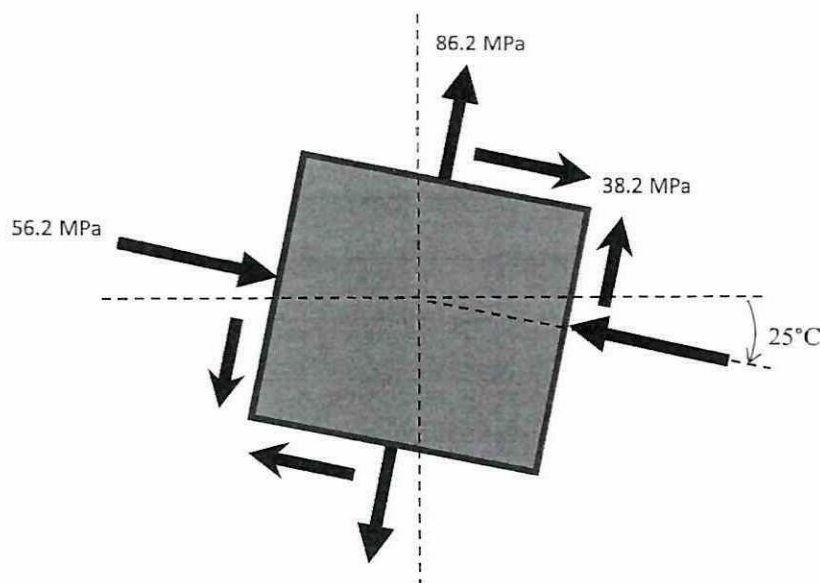
**Figure Q1.1**

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**PART B - ANSWER THREE (3) QUESTIONS ONLY**

**Q2** Answer the following questions.

- (a) External axial forces applied to a rigid body tend to deform the body. The external forces acting on a rigid body are termed loads, the deformation is known as strain, and the internal resistance to the external force is called stress. Sketch a stress-strain diagram for steel material including all its important characteristics. Then briefly discuss on the elastic limit and the yield point of the material. (5 marks)
- (b) A control rod made from a yellow brass must not stretch more than  $t$  mm when the tension force of  $P$  kN is applied. If the modulus of elasticity is  $E$  GPa and the maximum allowable normal stress is  $\sigma_{all}$  MPa, determine,
  - (i) The smallest allowable diameter for a yellow brass control rod. (3 marks)
  - (ii) The maximum length of the rod under these specified conditions. (2 marks)
- (c) **Figure Q2.1** shows the principal stress after the element has been rotated through  $25^\circ$  clockwise. Using the equation method, determine the stress component exerted on the axis before the rotation. (10 marks)



**Figure Q2.1**

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**Q3** Figure Q3.1 shows a steel beam is subjected to uniformly distributed load. A rigid bar DEF is connected at point D.

- (a) Determine support reaction at A and B. (6 marks)
- (b) Calculate deflection at point C using MacCaulay method. Take modulus of elasticity, E as  $5 \times 10^6 \text{ N/mm}^2$  and moment of inertia, I as  $9 \times 10^6 \text{ mm}^4$ . (14 marks)

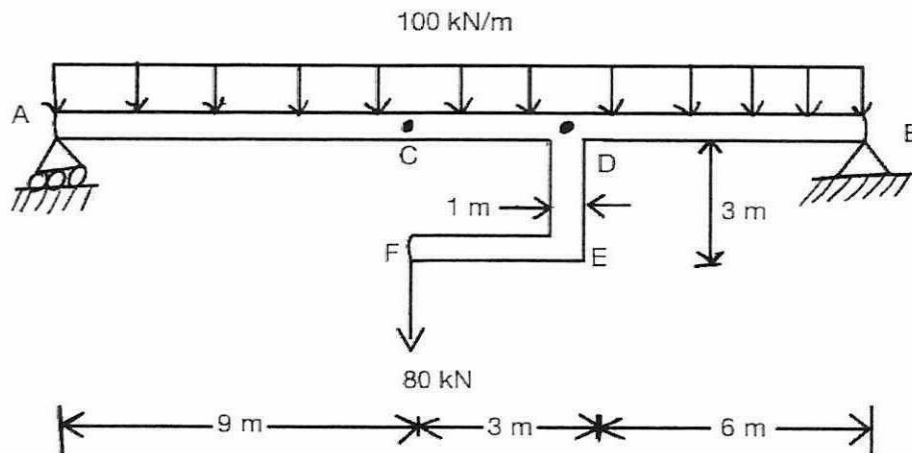


Figure Q3.1

**Q4** Answer the following questions.

- (a) The value of the stress corresponding to the critical load is called the critical stress and is denoted by  $\sigma_{cr}$ . Explain in detail the condition of column based on equation below subjected to Euler's formula.

$$\sigma_{cr} = \frac{\pi^2 E}{(L/r)^2}$$

(3 marks)

- (b) A column with a  $50 \text{ mm}^2$  section and 0.7 m effective length is made of the aluminium alloy 2014-T6. Determine the maximum load P for the column with an eccentricity of 20 mm. (7 marks)
- (c) Using the aluminium alloy 2014-T6 as shown in Figure Q4.1, determine the smallest diameter rod which can be used to support the centric load  $P = 60 \text{ kN}$  by assuming  $L/r > 55$  or  $L/r < 55$  if

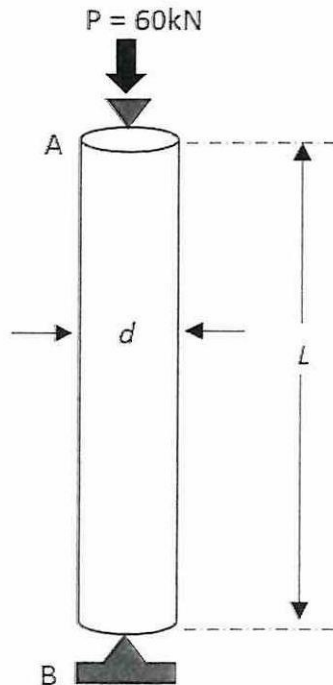
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(i)  $L = 750 \text{ mm}$

(5 marks)

(ii)  $L = 300 \text{ mm}$

(5 marks)



**Figure Q4.1**

**Q5** Answer the following questions.

(a) When deriving a simple theory to describe the behaviors of a uniform circular shaft subjected to torques, it is necessary to make a few basic assumptions. State **TWO (2)** of these assumptions.

(4 marks)

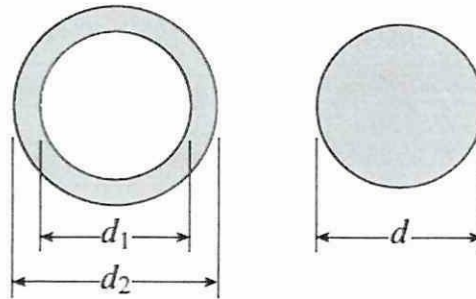
(b) **Figure Q5.1** shows an aluminium hollow pole used in a roof structure with an outside and inside diameter of  $d_2=104 \text{ mm}$  and  $d_1=82 \text{ mm}$ , respectively. The aluminium hollow pole has length of  $2.75 \text{ m}$  and shear modulus,  $G = 28 \text{ GPa}$ .

(i) If the aluminium hollow pole is twisted in pure torsion by torques acting at the ends, determine the angle of twist (in degrees) when the maximum shear stress is  $48 \text{ MPa}$ .

(8 marks)

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- (ii) Assuming aluminium solid pole is used, calculate the required diameter  $d$ . The pole need to resist the same torque with the same maximum stress = 48 MPa. (6 marks)
- (iii) Determine the ratio of the weight of the the aluminium hollow pole to the weight of the aluminium solid pole. (2 marks)



**Figure Q5.1**

**- END OF QUESTIONS-**

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APPENDIX A

LIST OF EQUATIONS

A. Stress and Strain

$$\sigma = \frac{P}{A} \quad \tau = \frac{P}{A} \quad \varepsilon = \frac{\delta}{L} \quad \delta = \frac{PL}{AE} \quad F.S = \frac{F_{fail}}{F_{allow}}$$

B. Stress transformation.

$$\sigma_x' = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\theta + \tau_{xy} \sin 2\theta$$

$$\sigma_y' = \frac{\sigma_x + \sigma_y}{2} - \frac{\sigma_x - \sigma_y}{2} \cos 2\theta - \tau_{xy} \sin 2\theta$$

$$\tau_{xy}' = -\frac{\sigma_x - \sigma_y}{2} \sin 2\theta + \tau_{xy} \cos 2\theta$$

$$R = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\sigma_{max,min} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2}$$

$$\tan 2\theta_p = \frac{2\tau_{xy}}{\sigma_x - \sigma_y}$$

C. Bending stress and shear stress

$$\frac{M}{I} = \frac{\sigma}{y} \quad \tau = \frac{VA\bar{y}}{It} \quad \tau_{max} = \frac{1.5V}{A}$$

D. Deformation of flexural member

(1)  $EI \frac{d^4 y}{dx^4} = q(x)$  Force-deflection equation

(2)  $EI \frac{d^3 y}{dx^3} = V(x) = \int q(x) dx + C_1$  Shear-deflection equation

(3)  $EI \frac{d^2 y}{dx^2} = M(x) = \iint q(x) dx^2 + C_1 x + C_2$  Bending moment-deflection equation

(4)  $EI \frac{dy}{dx} = EI\theta(x) = \int M(x) dx + C_3$  Slope-deflection equation  
 $= \iiint q(x) dx^3 + C_1 x^2 + C_2 x + C_3$

(5)  $EIy(x) = \int M(x) dx^2 + C_3 x + C_4$  Deflection equation  
 $= \iiint \int q(x) dx^4 + C_1 x^3 + C_2 x^2 + C_3 x + C_4$

E. Buckling of compression member

$$r = \sqrt{\frac{I}{A}}$$

$$\text{Slenderness ratio} = \frac{L}{r}$$

$$\sigma_{max} = \frac{P}{A} + \frac{M_{max}y}{I}$$

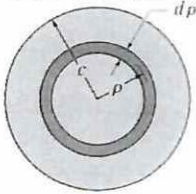
$$P_{cr} = \frac{\pi^2 EI}{L^2}$$

F. Torsion

$$\tau = \frac{Tc}{J}$$

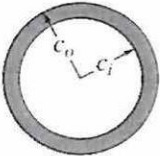
$$\phi = \frac{TL}{JG}$$

Solid Section



$$J = \frac{\pi}{2} c^4$$

Hollow Section



$$J = \frac{\pi}{2} c_o^4 - c_i^4$$

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